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APPLICATION NO.	FILING DATE	FIRST NAMED INVENTOR	ATTORNEY DOCKET NO.	CONFIRMATION NO.
10/576,663	02/22/2007	Quentin Roberts	H0005838-2930	8172
HONEYWELL TURBO TECHNOLOGIES 3201 WEST LOMITA BOULEVARD (LAW DEPARTMENT)			EXAMINER	
			PRAGER, JESSE M	
TORRANCE, CA 90505		ART UNIT	PAPER NUMBER	
			3745	
			MAIL DATE	DELIVERY MODE
			01/19/2010	PAPER

Please find below and/or attached an Office communication concerning this application or proceeding.

The time period for reply, if any, is set in the attached communication.

	Application No.	Applicant(s)				
Office Action Comment	10/576,663	ROBERTS ET AL.				
Office Action Summary	Examiner	Art Unit				
	JESSE PRAGER	3745				
The MAILING DATE of this communication appears on the cover sheet with the correspondence address Period for Reply						
A SHORTENED STATUTORY PERIOD FOR REPLY IS SET TO EXPIRE 3 MONTH(S) OR THIRTY (30) DAYS, WHICHEVER IS LONGER, FROM THE MAILING DATE OF THIS COMMUNICATION. - Extensions of time may be available under the provisions of 37 CFR 1.136(a). In no event, however, may a reply be timely filed after SIX (6) MONTHS from the mailing date of this communication. - If NO period for reply is specified above, the maximum statutory period will apply and will expire SIX (6) MONTHS from the mailing date of this communication. - Failure to reply within the set or extended period for reply will, by statute, cause the application to become ABANDONED (35 U.S.C. § 133). Any reply received by the Office later than three months after the mailing date of this communication, even if timely filed, may reduce any earned patent term adjustment. See 37 CFR 1.704(b).						
Status						
1) Responsive to communication(s) filed on 12/07	7/2009					
· <u> </u>	/					
	Since this application is in condition for allowance except for formal matters, prosecution as to the merits is closed in accordance with the practice under <i>Ex parte Quayle</i> , 1935 C.D. 11, 453 O.G. 213.					
closed in accordance with the practice under Ex pane Quayle, 1935 C.D. 11, 455 O.G. 215.						
Disposition of Claims						
4)⊠ Claim(s) <u>15, 17-19 and 21-22</u> is/are pending in	the application.					
	4a) Of the above claim(s) is/are withdrawn from consideration.					
5) Claim(s) is/are allowed.						
6)⊠ Claim(s) <u>15, 17-19 and 21-22</u> is/are rejected.						
7) Claim(s) is/are objected to.						
·						
8) Claim(s) are subject to restriction and/or election requirement.						
Application Papers						
9)☐ The specification is objected to by the Examiner.						
10)⊠ The drawing(s) filed on <u>21 April 2006</u> is/are: a)⊠ accepted or b)⊡ objected to by the Examiner.						
Applicant may not request that any objection to the drawing(s) be held in abeyance. See 37 CFR 1.85(a).						
Replacement drawing sheet(s) including the correction is required if the drawing(s) is objected to. See 37 CFR 1.121(d).						
11) The oath or declaration is objected to by the Examiner. Note the attached Office Action or form PTO-152.						
Priority under 35 U.S.C. § 119						
	12) Acknowledgment is made of a claim for foreign priority under 35 U.S.C. § 119(a)-(d) or (f).					
a) ☐ All b) ☐ Some * c) ☐ None of:						
	1. Certified copies of the priority documents have been received.					
2. Certified copies of the priority documents have been received in Application No						
3. Copies of the certified copies of the priority documents have been received in this National Stage						
application from the International Bureau (PCT Rule 17.2(a)).						
* See the attached detailed Office action for a list of the certified copies not received.						
• · · · · · · · · · · · · · · · · · · ·						
Attachment(s)						
Notice of References Cited (PTO-892) Notice of Draftsperson's Patent Drawing Review (PTO-948)	4)					
3) Information Disclosure Statement(s) (PTO/SB/08)	5) Notice of Informal Pa					
Paper No(s)/Mail Date 6) Other:						

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DETAILED ACTION

Claims 15, 17-19, and 21-22 are pending. Claims 16, 20, and 23-26 have been canceled.

Response to Arguments

1. Applicant's arguments, see pg. 5, filed 12/07/2009, with respect to the rejection(s) of claim(s) 16 and 20 under 35 U.S.C. 103(a) as being unpatentable over Erdmann (WO 2003/080999) in view of Leavesley (WO 2001/0053679) have been fully considered and are persuasive. In particular applicant argues, the combination of the references would not support one of ordinary skill arranging the slots to extend radially inwardly from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston. Therefore, the rejection has been withdrawn. However, upon further consideration, a new ground(s) of rejection is made in view of Leavesley (WO 03/023194). Leavesley (WO 03/023194) discloses slots extending from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston (Fig. 11).

Claim Rejections - 35 USC § 103

2. Claims 15, 17, 19 and 21 are rejected under 35 U.S.C. 103(a) as being unpatentable over Erdmann et al. (WO 2003/080999) in view of Leavesley (US Patent 5,231,831) and in further view of Leavesley (WO 03/023194).

In regards to claim 15, Erdmann et al. discloses a turbine housing assembly for an exhaust-gas-driven turbocharger, comprising:

a sector-divided turbine housing (Fig. 2) defining a generally annular chamber structured and arranged to surround a turbine wheel (3), a plurality of circumferentially spaced dividing walls (25, 26) extending generally radially inwardly from a radially outer wall of the chamber

and dividing the chamber into a plurality of separate angular sectors each of which occupies a fractional part of a circumference of the chamber, each sector of the chamber at a radially inward side thereof having an axial length;

a vane assembly (10, 23 and 24) for guiding flow from the chamber into the turbine wheel, the vane assembly comprising a ring of circumferentially spaced vanes that include dividing vanes (23, 24) and additional vanes (10) mounted on a fixed structure of the turbine housing assembly, the dividing vanes corresponding in number to the number of dividing walls, each dividing vane forming an extension of one of the dividing walls and extending generally radially inwardly from the dividing wall and terminating at a trailing edge of the dividing vane, the additional vanes being located circumferentially between the dividing vanes; and

a variable-geometry mechanism (Fig. 1) comprising a tubular piston (17) disposed radially inward of the chamber and axially slidable relative to the chamber between a fully open position a closed position in which a fractional portion of the axial length of the sectors is blocked by the piston, the piston having a radially outer surface and a radially inner surface defining a radial thickness of the piston therebetween, wherein the piston and the vanes overlap radially and at least the dividing vanes are recieved in axially extending slots in the piston when the piston is in the closed postion;

wherein the dividing vanes (23,24) extend fully across the axial length of the sectors so that the sector-division of the turbine housing is preserved when the piston is in the fully open position.

Erdmann et al. is silent as to whether

the slots extend radially inwardly from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston, such that the slots do not go all the way through to the radially inner surface of the piston; and

the additional vanes extend along less than the axial length of the sectors and are axially located such that when the piston is in the closed position a portion of the sectors remain open and the additional vanes extend fully across the portion, and when the piston is in the fully open position there is a space between ends of the additional vanes and an end of the piston.

Leavesley (Patent '831) discloses additional vanes (102) extend along less than the axial length of the sectors (104) and are axially located such that when the piston is in the closed position a portion of the sectors remain open and the additional vanes extend fully across the portion (Fig. 14), and when the piston is in the fully open position there is a space between ends of the additional vanes and an end of the piston (Fig. 15).

It would have been obvious to one having ordinary skill in the art at the time the invention was made to modify the turbine housing assembly of Erdmann et al. by providing additional vanes axially shorter than the sectors, the additional vane structure extending across the portion and a portion of the sectors remaining open in the closed position, and having a space between ends of the additional vanes and an end of the piston in the open position, as taught by Leavesley because the overlapping design of the vanes of Leavesley enables a more appropriately responsive turbocharger (Col. 1, lines 25-41).

Leavesley (WO 03/023194) discloses the slots extend radially inwardly from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston, such that the slots do not go all the way through to the radially inner surface of the piston (Fig. 11).

It would have been obvious to one having ordinary skill in the art at the time the invention was made to further modify the turbine housing assembly of Erdmann et al. by cutting the slots so that they extend radially inwardly from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston, such that the slots do not go all the way through to the radially inner surface of the piston, as taught by Leavesley (WO 03/023194), as an economically means of producing the slots in the piston (Leavesley Pg 14, lines 1-3).

In regards to claim 17, the modified turbine housing assembly of Erdmann et al. comprises the additional vanes and piston are arranged such that in the closed position of the piston, an end of the piston abuts ends of the additional vanes.

In regards to claim 19, Erdmann et al. discloses a turbine for an exhaust-gas-driven turbocharger, comprising:

a turbine wheel (3);

a sector-divided turbine housing (Fig. 2) defining a generally annular chamber structured and arranged to surround the turbine wheel (3), a plurality of circumferentially spaced dividing walls (25, 26) extending generally radially inwardly from a radially outer wall of the chamber and dividing the chamber into a plurality of separate angular sectors each of which occupies a fractional part of a circumference of the chamber, each sector of the chamber at a radially inward side thereof having an axial length;

a vane assembly (10, 23 and 24) for guiding flow from the chamber into the turbine wheel, the vane assembly comprising a ring of circumferentially spaced vanes that include dividing vanes (23, 24) and additional vanes (10) mounted on a fixed structure of the turbine housing assembly, the dividing vanes corresponding in number to the number of dividing walls,

each dividing vane forming an extension of one of the dividing walls and extending generally radially inwardly from the dividing wall and terminating at a trailing edge of the dividing vane, the additional vanes being located circumferentially between the dividing vanes; and

a variable-geometry mechanism (Fig. 1) comprising a tubular piston (17) disposed radially inward of the chamber and axially slidable relative to the chamber between a fully open position a closed position in which a fractional portion of the axial length of the sectors is blocked by the piston, the piston having a radially outer surface and a radially inner surface defining a radial thickness of the piston therebetween, wherein the piston and the vanes overlap radially and at least the dividing vanes are recieved in axially extending slots in the piston when the piston is in the closed postion;

wherein the dividing vanes (23,24) extend fully across the axial length of the sectors so that the sector-division of the turbine housing is preserved when the piston is in the fully open position.

Erdmann et al. is silent as to whether

the slots extend radially inwardly from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston, such that the slots do not go all the way through to the radially inner surface of the piston; and

the additional vanes extend along less than the axial length of the sectors and are axially located such that when the piston is in the closed position a portion of the sectors remain open and the additional vanes extend fully across the portion, and when the piston is in the fully open position there is a space between ends of the additional vanes and an end of the piston.

Leavesley (Patent '831) discloses additional vanes (102) extend along less than the axial length of the sectors (104) and are axially located such that when the piston is in the closed position a portion of the sectors remain open and the additional vanes extend fully across the portion (Fig. 14), and when the piston is in the fully open position there is a space between ends of the additional vanes and an end of the piston (Fig. 15).

It would have been obvious to one having ordinary skill in the art at the time the invention was made to modify the turbine of Erdmann et al. by providing additional vanes axially shorter than the sectors, the additional vane structure extending across the portion and a portion of the sectors remaining open in the closed position, and having a space between ends of the additional vanes and an end of the piston in the open position, as taught by Leavesley because the overlapping design of the vanes of Leavesley enables a more appropriately responsive turbocharger (Col. 1, lines 25-41).

Leavesley (WO 03/023194) discloses the slots extend radially inwardly from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston, such that the slots do not go all the way through to the radially inner surface of the piston (Fig. 11).

It would have been obvious to one having ordinary skill in the art at the time the invention was made to further modify the turbine of Erdmann et al. by cutting the slots so that they extend radially inwardly from the radially outer surface of the piston for a radial distance less than the radial thickness of the piston, such that the slots do not go all the way through to the radially inner surface of the piston, as taught by Leavesley (WO 03/023194), as an economically means of producing the slots in the piston (Leavesley Pg 14, lines 1-3).

In regards to claim 21, the modified turbine housing assembly of Erdmann et al. comprises the additional vanes and piston are arranged such that in the closed position of the piston, an end of the piston abuts ends of the additional vanes.

3. Claims 18 and 22 are rejected under 35 U.S.C. 103(a) as being unpatentable over Erdmann et al. (WO 2003/080999) in view of Leavesley (US Patent 5,231,831) and in view of Leavesley (WO 03/023194), as applied to claims 15 and 19, and in further view of Bernardini et al. (WO 01/053679).

The modified Erdmann et al. turbine housing and turbine contains all of the claim limitations except the dividing vanes and additional vanes are mounted on a ring-shaped member separate from the turbine housing.

Bernardini et al. discloses the vanes mounted on a ring-shaped member separate from the turbine housing (Fig. 1).

It would have been obvious to one having ordinary skill in the art at the time the invention was made to further modify the Erdmann et al. turbocharger such that the vanes are mounted on a ring-shaped member separate on the piston separate from the turbine housing, as taught by Bernardini et al., because the vanes on the piston are protected from thermal stresses by the heat screen.

Any inquiry concerning this communication or earlier communications from the examiner should be directed to JESSE PRAGER whose telephone number is (571)270-1412. The examiner can normally be reached on Monday-Friday, 9:00 am - 5:00 pm, EST.

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If attempts to reach the examiner by telephone are unsuccessful, the examiner's supervisor, Edward Look can be reached on (571)272-4820. The fax phone number for the organization where this application or proceeding is assigned is 571-273-8300.

Information regarding the status of an application may be obtained from the Patent Application Information Retrieval (PAIR) system. Status information for published applications may be obtained from either Private PAIR or Public PAIR. Status information for unpublished applications is available through Private PAIR only. For more information about the PAIR system, see http://pair-direct.uspto.gov. Should you have questions on access to the Private PAIR system, contact the Electronic Business Center (EBC) at 866-217-9197 (toll-free). If you would like assistance from a USPTO Customer Service Representative or access to the automated information system, call 800-786-9199 (IN USA OR CANADA) or 571-272-1000.

/JESSE PRAGER/ Examiner, Art Unit 3745

1/11/2010

/Edward K. Look/ Supervisory Patent Examiner, Art Unit 3745